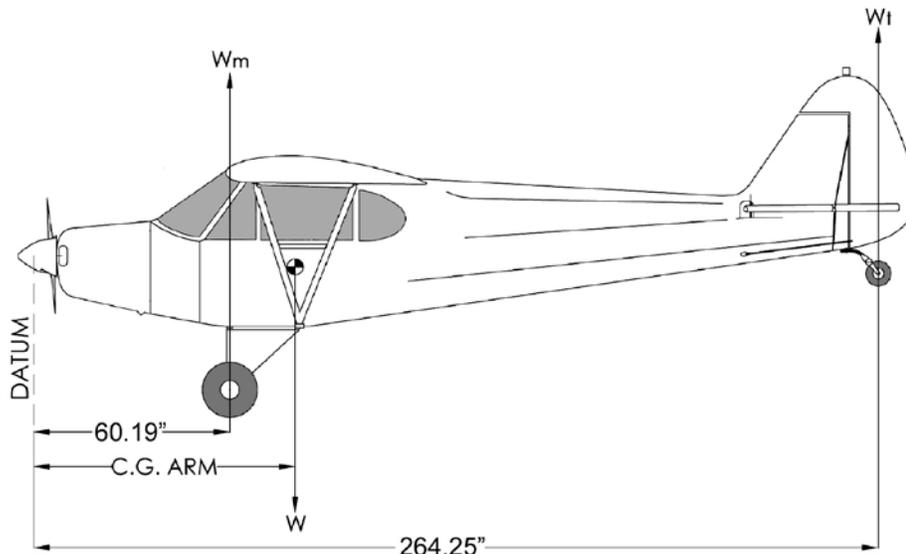


- Once the aircraft has been leveled, record the weight on the main wheels and the tailwheel.



Aircraft Geometry

$$CG\ Arm = \frac{(W_m \times 60.19) + (W_t \times 264.25)}{W}$$

- CG Arm = Distance from the datum to the center of gravity (in inches)
- W = Total weight of the aircraft
- W<sub>m</sub> = Sum of the weight on both main wheels
- W<sub>t</sub> = Weight on the tailwheel

Aircraft was weighed with no useable fuel, all equipment installed, and 5 quarts of oil installed.

Weight as follows:

Left Main	Tailwheel	Right Main
<u>478lbs.</u>	<u>45lbs.</u>	<u>478lbs.</u>

10lbs of ballast was added at station 236.6" after weighing.

NXXXXX

WEIGHT AND BALANCE

CCK-1865-000X

Make: XXXXXXXXXXXX Model: CARBON CUB EX

Date: XX/XX/XXXX

### PERTINENT INFORMATION FOR WEIGHT AND BALANCE

Position of datum 60 inches ahead of wing leading edge

Maximum gross weight ..... 1865 lbs

Center of gravity limits at 1865 lb.

Forward ..... 74.0 in. aft of datum

Aft ..... 79.0 in. aft of datum

Center of gravity limits at 1600 lb. or less

Forward ..... 70.5 in. aft of datum

Aft ..... 79.0 in. aft of datum

Make: XXXXXXXXXXXXXXXXXXXX

Model: Carbon Cub EX

Aircraft serial number: CCK-1865-000X

Registration number: NXXXX

Date: XX/XX/XXX

Basic empty weight (lbs)	Arm (inches)	Moment (in-lbs)
1001lbs.	71.02	71091.0lbs.

### Basic Empty Weight

Useful load = Maximum takeoff weight- Basic empty weight

= 1865 – 1001lbs

Useful Load = 864lbs

NXXXXX

WEIGHT AND BALANCE

CCK-1865-000X

Make: XXXXXXXXXXXX Model: CARBON CUB EX

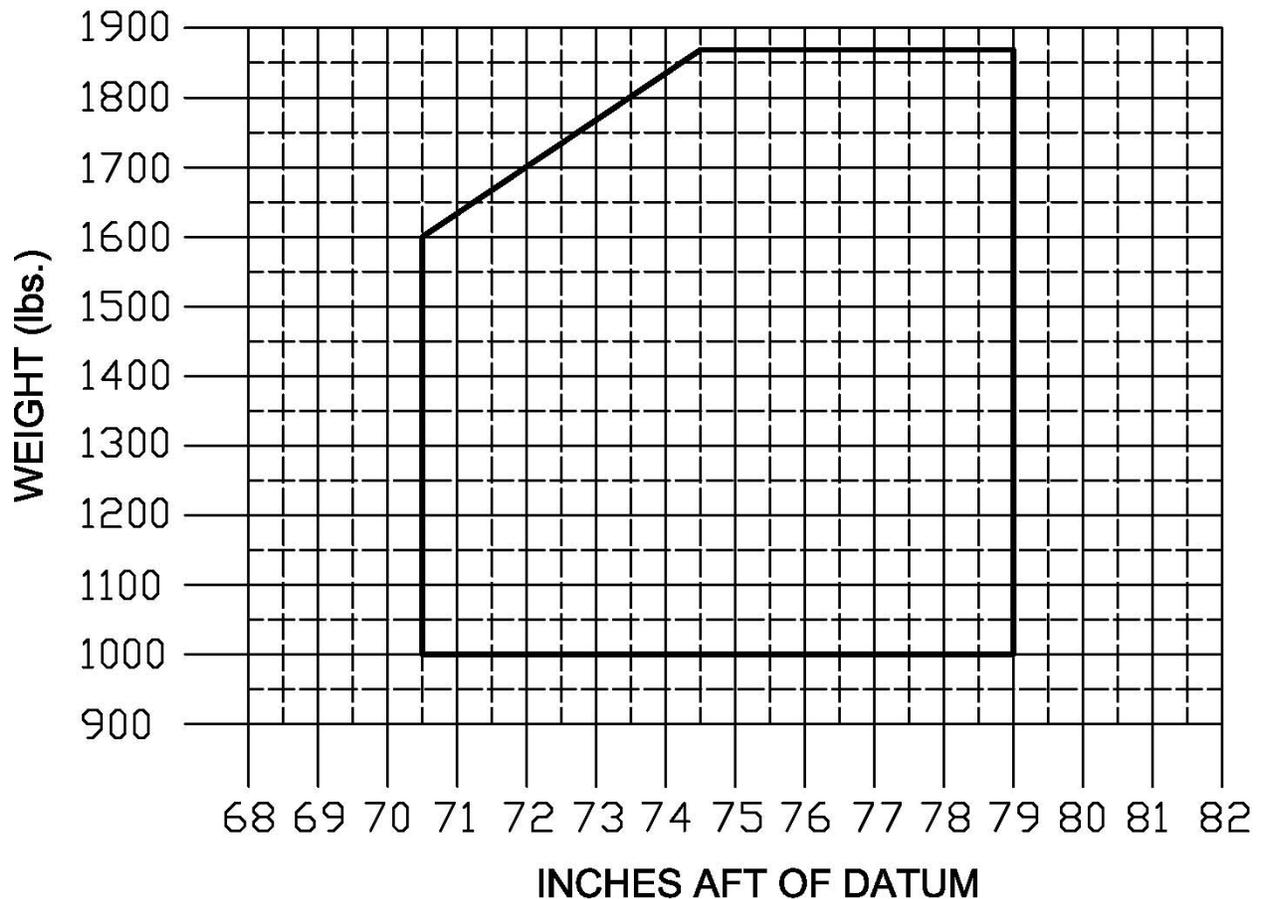
Date: XX/XX/XXXX

*SAMPLE WEIGHT AND BALANCE CALCULATION*

**WEIGHT AND BALANCE DETERMINATION FOR FLIGHT**

In order to calculate the weight and balance of the aircraft:

1. Insert the respective loads in table below
2. Multiply each load by its respective arm and note the moment.
3. Add the loads to calculate the takeoff weight
4. Add the moments to compute the total moment.
5. Divide the moment by the takeoff weight. This is the final position of the center of gravity.
6. Plot the point on the table below. If it is within the weight and balance envelope, the aircraft is within the approved envelope.



NXXXXX

WEIGHT AND BALANCE

CCK-1865-000X

Make: XXXXXXXXXXXX Model: CARBON CUB EX

Date: XX/XX/XXXX

Item	Weight (lbs)	Moment Arm (in)	Moment (in-lbs)
Aircraft Empty	1001	71.02	71091.0
Fuel	110	83.90	9229.0
Pilot	170	71.40	12138.0
Passenger	200	96.08	19216.0
Forward Cargo Compartment (100 lb max)	100	110.40	11040.0
Extended Cargo Compartment (60 lb max)	40	134.40	5376.0
<b>TOTAL</b>	<b>1621</b>		<b>128090.0</b>
<b>CG Station:</b>	<b>79.0</b>		

Item	Weight (lbs)	Moment Arm (in)	Moment (in-lbs)
Aircraft Empty	1001	71.02	71091.0
Fuel	254	83.90	21310.6
Pilot	250	71.40	17850.0
Passenger	180	96.08	17294.4
Forward Cargo Compartment (100 lb max)	100	110.40	11040.0
Extended Cargo Compartment (60 lb max)	40	134.40	5376.0
<b>TOTAL</b>	<b>1825</b>		<b>143962.0</b>
<b>CG Station:</b>	<b>78.9</b>		

Item	Weight (lbs)	Moment Arm (in)	Moment (in-lbs)
Aircraft Empty	1001	71.02	71091.0
Fuel	254	83.90	21310.6
Pilot	275	71.40	19635.0
Passenger	0	96.08	0.0
Forward Cargo Compartment (100 lb max)	100	110.40	11040.0
Extended Cargo Compartment (60 lb max)	60	134.40	8064.0
<b>TOTAL</b>	<b>1690</b>		<b>131140.6</b>
<b>CG Station:</b>	<b>77.6</b>		